

# Experimental study of an active control for a faulty perturbed flexible structure

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**Abstract**—In this paper, a linear dynamic control is developed for a faulty flexible structure subject to external ground perturbation. This active controller is based on  $H_\infty$  theory and is designed using linear matrix inequality (LMI) theory. Lyapunov theory is invoked to validate the control design. According to experiments, where a two levels flexible building with active mass damper and external perturbation is employed, show that this strategy improves controller performance when it is compared with a given controller. This design faces the nonlinear structural system too. *20th Mediterranean Conference on Control and Automation (MED12)*.

## I. INTRODUCTION

The purpose of this paper is to design a linear robust controller able to attenuate vibrations (produced by the external ground perturbation) on a flexible structure using the active mass damper controller. This is conceptually similar to active mass damper being studied in earthquake mitigation research facilities to reduce damage from earthquakes on high rise buildings ([3], [9]). For the purpose of maintaining the seismic response of structures within safety levels, service and comfort limits, the combination of passive base isolators and feedback controllers has been proposed in recent years ([3], [2], [11], [14]).

For linear systems,  $H_\infty$  control theory offers the possibility of including robustness considerations explicitly in the design and the opportunity to formulate physically meaningful performance objectives that can be expressed as  $H_\infty$  design specifications. In the state-space formulation, the problem of reducing the  $H_\infty$  norm or, equivalently, its induced  $L_2$ -norm, of the closed-loop system is viewed as a solution of the algebraic Riccati inequalities arising in linear quadratic differential theory (see [8] for references on the  $H_\infty$  design). In order to solve these Riccati inequalities, they are expressed in terms of linear matrix inequalities (LMI) [10].

In this paper, the design of a local controller that stabilizes the two levels flexible structure, in the presence of external disturbances, is studied (see Figure 1). The main objective is to demonstrate how a solution to such a complex control problem can be obtained by applying the efficient tools of



Fig. 1. Quanser Shake Table II without structure fault.

LMI theory. To show the robustness of the proposed controller, laboratory experiments are carried on. The solution is presented using the robust  $H_\infty$  control theory [8] by the method of linear matrix inequality ([1], [5], [10]), that can be applied to structural control ([18], [4]). Our objective is to present an easy control algorithm that solves a complex nonlinear problem. Moreover, an advantage of the active control based on LMI techniques is the robustness to external disturbance, as it is well known, but also its simple design ([12], [16]). In this sense, the law is defined a priori, before doing the experiments, and it only depends on the linearized structural system. It is relatively easy to design the controller because it does not rely on online controller parameters adjustment.

An experimental shake table is used to simulate earthquakes exciting the flexible modes of a tall structure [17]. It is composed by a flexible two levels building with an active mass damper at the top of the building. Only acceleration measurements at level one and two are available, and the position measurement of the cart (the active mass damper) situated at the building top (see Figure 1). In [13], the authors present a solution where chattering control theory is used. Now, a simpler controller is designed and a modification of the experiment is introduced to study the performance in front of additional flexible devices and model uncertainties. Our faulty perturbed flexible structure experiment is as follows (see Figures 2-4). Floor one is disjointed from floor two, where a cylindric roller is placed between them. Then, elastic links are used to fix the floors one and two. Figure 2 pictures

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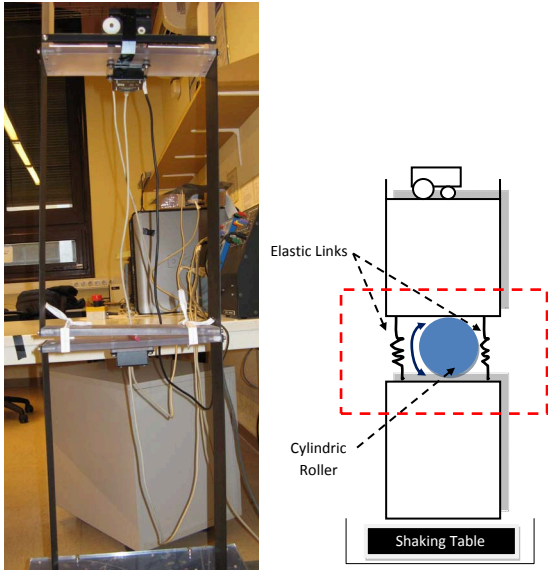


Fig. 2. Modification of Quanser Shake Table II (left: real experiment; right: scheme).

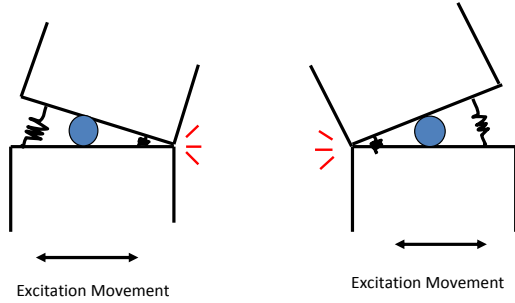


Fig. 3. Movement of shake table under ground excitation.

this faulty experimental modification. Experiments are done to study the effectiveness of the linear  $H_\infty$  dynamic controller (based on LMI techniques), when sinusoidal chirp external perturbation is present. This scenario can appear in structural control in transport industry (containerization). For instance, suppose that we have two stacked containers fixed together by elastic links. During the transportation, the top container can slide similarly to the present experiment. Figures 3 and 4 show the possible movements of the containers or levels under transportation excitation. Moreover, this fault is too similar to the dead-zone appearing in mechanical systems

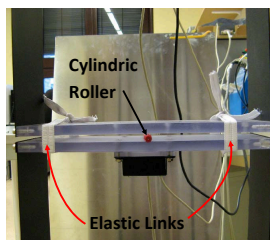


Fig. 4. Zoom of modified experiment presented in Figure 2.

with impacts [6].

The paper is structured as follows. Section II presents the linear description of the two-floors structure with a single active mass damper located at the top of the structure. The control objective is stated, displaying the dynamic  $H_\infty$  control solution in terms of LMI. In Section III experimental results are shown, proven the effectiveness of the methodology. Finally, conclusions and future work are stated in Section IV.

## II. CONTROL DESIGN

### A. Dynamic unfaulty structure model

This section provides the statement of the differential equation of motion for the two levels building. For small floor deflection angles, both floors are modeled as standard linear spring-mass systems [4]. In our approach, the structure viscous damping coefficients are neglected. The model is derived using Lagrangian formulation, where the dynamic equations are obtained and then a linear model is derived by linearizing about the quiescent (latent) point [17]. The states are defined as:

$$x^T(t) = [x_c(t) \ x_{f_1}(t) \ x_{f_2}(t) \ \dot{x}_c(t) \ \dot{x}_{f_1}(t) \ \dot{x}_{f_2}(t)], \quad (1)$$

where  $x_{f_i}(t)$  is the position of the floor  $i = 1, 2$  (floor deflection), and  $x_c(t)$  is the cart position (see Figure 1). The available measurements are  $x_c(t)$ ,  $\ddot{x}_{f_1}(t)$  and  $\ddot{x}_{f_2}(t)$ . That means  $y(t) = [x_c(t) \ , \ \ddot{x}_{f_1}(t) \ , \ \ddot{x}_{f_2}(t)]^T$  must be introduced as the measured output. The linear model about the quiescent point is defined by:

$$\begin{aligned} m_c m_{f_2} \ddot{x}_c - m_c k_{f_2} x_{f_2} &= (m_c + m_{f_2}) u(t) \\ m_{f_1} \ddot{x}_{f_1} + k_{f_1} x_{f_1} - k_{f_2} x_{f_2} &= 0 \\ m_{f_1} m_c m_{f_2} r_{mp}^2 \ddot{x}_{f_2} - m_{f_2} m_c m_{f_2} r_{mp}^2 k_{f_1} x_{f_1} &+ \\ &+ (m_{f_1} m_c + m_{f_2} m_c) r_{mp}^2 k_{f_2} x_{f_2} = -m_{f_1} m_c r_{mp}^2 u(t) \end{aligned} \quad (2)$$

where  $m_c$ ,  $m_{f_1}$  and  $m_{f_2}$  are the cart, first and second floor mass;  $k_{f_1}$  and  $k_{f_2}$  are the linear stiffness constants of both floors;  $r_{mp}$  is the cart motor pinion radius; and the driving force of the motor cart is  $u(t)$  (control input). Taking into account (2), the state space representation results to be:

$$\begin{aligned} \dot{x}(t) &= Ax(t) + Bu(t) \\ y(t) &= Cx(t) + Du(t). \end{aligned} \quad (3)$$

The matrices are defined as [17]:

$$A = \begin{bmatrix} 0 & 0 & 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 \\ 0 & 0 & 278.43 & -18.69 & 0 & 0 \\ 0 & -431.03 & 431.03 & 0 & 0 & 0 \\ 0 & 431.03 & -766.49 & 5.98 & 0 & 0 \end{bmatrix},$$

$$B = [0 \ 0 \ 0 \ 3.01 \ 0 \ -0.96]^T,$$

$$C = \begin{bmatrix} 1 & 0 & 0 & 0 & 0 & 0 \\ 0 & -431.03 & 431.03 & 0 & 0 & 0 \\ 0 & 431.03 & -766.49 & 5.98 & 0 & 0 \end{bmatrix},$$

$$D = \begin{bmatrix} 0 \\ 0 \\ -0.96 \end{bmatrix}. \quad (4)$$

### B. Linear dynamic control

In the structure, the active damper is located at the top of the building. For this reason, we want to study the effect of external disturbance  $w(t) \in \mathcal{L}_2$  on the cart position  $x_c(t)$  and floor two acceleration  $\ddot{x}_{f_2}(t)$ , where the cart is located. Therefore, a performance variable (virtual output)  $z(t)$  is defined as  $z^T(t) = [x_c(t) \ \ddot{x}_{f_2}(t) \ u(t)]$ . The state-space representation of system (3)-(4) yields

$$\begin{cases} \dot{x}(t) = Ax(t) + Bu(t) + B_1w(t), \\ z(t) = C_1x(t) + D_{12}u(t), \\ y(t) = Cx(t) + Du(t) + D_{21}w(t), \end{cases} \quad (5)$$

where

$$B_1 = [0 \ 1 \ 0 \ 0 \ 0 \ 0],$$

$$C_1 = \begin{bmatrix} 100 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 \\ 0 & 0 & 0 & 0 & 0 & 0 \end{bmatrix},$$

$$D_{12} = [0 \ 0 \ 0.1]^T \text{ and } D_{21} = [0 \ 0 \ 1]^T.$$

Matrix  $B_1$  is defined to take into account that the external perturbation  $w(t)$  is produced on the ground. Matrices  $C_1$  and  $D_{12}$  are defined to increase the weight of the cart position (where the controller is located) in front of the external perturbation. The above dynamic model satisfies the following standard  $H_\infty$  assumptions<sup>1</sup> [8]:

- 1)  $(A, B, C)$  is stabilizable and detectable;
- 2)  $(A, B_1, C_1)$  is stabilizable and detectable;
- 3)  $D_{12}^T(C_1 D_{12}) = (0 \ \alpha)$  with  $\alpha > 0$ ;
- 4)  $\begin{bmatrix} B_1 \\ D_{21} \end{bmatrix} D_{21}^T = \begin{bmatrix} 0 \\ I \end{bmatrix}$ .

The problem of robust controller with guaranteed  $H_\infty$  performance is addressed to answer this question: Does there exist a feedback control such that the  $H_\infty$  norm of the closed-loop system from input disturbance  $w(t) \in \mathcal{L}_2$  to output  $z(t)$  is less than some prescribed value  $\gamma \in \mathbf{R}^+$ ? In order to solve this problem, the LMI techniques are used.

<sup>1</sup>In [8], it is assumed that  $D = 0$ . A transformation from  $D$  nonzero to  $D = 0$  is explained in [15].

Let's consider an  $H_\infty$  controller  $u(t)$  designed as a dynamic control strictly proper [1]:

$$K: \begin{cases} \dot{\phi}(t) = A_k \phi(t) + B_k y(t) \\ u(t) = C_k \phi(t). \end{cases} \quad (6)$$

Considering the variable  $\eta(t)^T = (x(t)^T, \phi(t)^T)$ , the closed loop system (5)-(6) is defined by:

$$\dot{\eta}(t) = \underbrace{\begin{bmatrix} A & BC_k \\ B_k C & A_k + B_k DC_k \end{bmatrix}}_{\mathcal{A}} \eta(t) + \underbrace{\begin{bmatrix} B_1 \\ B_k D_{21} \end{bmatrix}}_{\mathcal{B}} w(t),$$

$$z(t) = \underbrace{(C_1 \ D_{12} C_k)}_{\mathcal{C}_\infty} \eta(t),$$

$$y(t) = \underbrace{(C \ DC_k)}_{\mathcal{C}_{2\infty}} \eta(t) + D_{12} w(t). \quad (7)$$

Our first control objective is to design a dynamic control (6) such that it stabilizes the closed loop system (7) under  $\mathcal{L}_2$  disturbances, employing  $H_\infty$ -LMI theory. A practical way to solve this problem is to consider a Lyapunov function  $V_1(t)$  such that for any nonzero  $\eta(t)$  and input  $w(t) \in \mathcal{L}_2$ , the following condition holds (see [1] for details):

$$\frac{d}{dt} V_1(t) + \gamma^{-1} z^T(t) z(t) - \gamma w^T(t) w(t) < 0. \quad (8)$$

Then, an  $H_\infty$  performance bound for the closed-loop system (7) is ensured.

*Definition 1 ( $H_\infty$  Controller [1]):* If there exists a matrices  $A_k$ ,  $B_k$  and  $C_k$  such that (8) holds, the control law  $u(t)$  (6) is said to be an  $H_\infty$  controller for the system (5), that is, the system is internally stable<sup>2</sup> with  $H_\infty$  norm less than  $\gamma$ , i.e.,  $\|z\|_\infty \leq \gamma^2 \|w\|_\infty$  for  $w \in \mathcal{L}_2$ .

The  $H_\infty$  characterization of the robust control begins by considering the next Lyapunov function:

$$V_1(t) = \eta(t)^T P \eta(t), \quad (9)$$

with  $P > 0$ . Imposing the  $H_\infty$  condition (8), we obtain:

$$\begin{aligned} (\mathcal{A} \eta + \mathcal{B} w) P \eta + \eta^T P^T (\mathcal{A} \eta + \mathcal{B} w)^T + \\ + \gamma^{-1} \eta^T \mathcal{C}_\infty^T \mathcal{C}_\infty \eta - \gamma w^T w < 0. \end{aligned} \quad (10)$$

We can rewrite inequality (10) as a matrix inequality by applying the Schur complement [1]:

$$\begin{bmatrix} \mathcal{A}^T P + P \mathcal{A} & P \mathcal{B} & \mathcal{C}_\infty^T \\ \mathcal{B}^T P & -\gamma & 0 \\ \mathcal{C}_\infty & 0 & -\gamma \end{bmatrix} < 0. \quad (11)$$

This matrix inequality is feasible and solvable numerically using LMI-techniques included in the robust control toolbox

<sup>2</sup>Internally stable means that the closed-loop system is asymptotically stable when  $w = 0$ .

of Matlab [7]. Given the continuous-time plant (5), the sub-optimal  $H_\infty$  performance  $\gamma$  is computed, as well as an  $H_\infty$  controller  $K$  (6) that internally stabilizes the plant, yielding a closed-loop gain no larger than  $\gamma$ .

### III. EXPERIMENTS: TESTING ROBUSTNESS

#### A. Experimental setup

Shake Table II is an instrumental shake table developed by Quanser Inc. [17]. The system is comprised by a shake table, a universal power module, a data acquisition card (DAC) along with its external terminal board, and a PC running control software. The PC sends and receives signals through the DAC using the WinCon software. Designed to simulate earthquakes and evaluate the performance of active mass damper, the shake table consists of an 1 Hp brushless servo motor driving a lead screw ([17], [4]). The lead screw drives a circulating ball nut which is coupled to the 18"  $\times$  18" table (see Figure 5). The table itself slides on a low friction linear ball bearings on 2 ground-hardened shafts. It can drive a 15 Kg. mass at 2.5 g. Maximum travel is  $\pm 7$ cm. In this paper, the external disturbance considered is a sine chirp wave (Quanser Chirp input  $w(t)$  in Figure 6) with increasing frequency from 0.1Hz to 0.7Hz and target time 40s, the total time of the experiments.

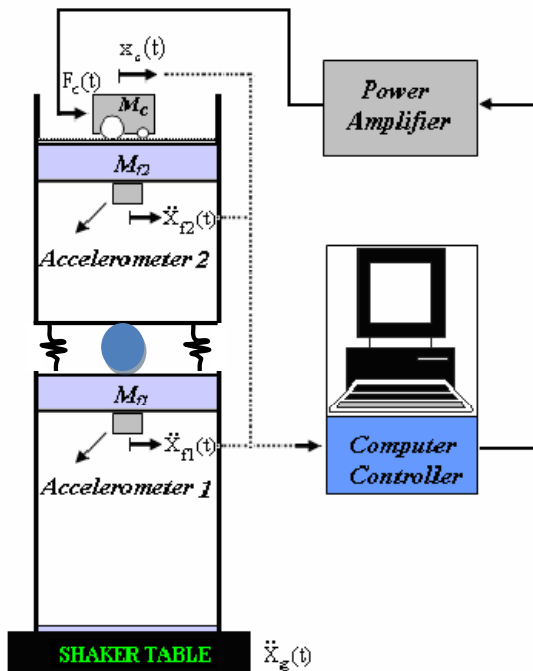


Fig. 5. Experimental set-up.

The shake table operating simulink program is presented in Figure 6, where the proposed dynamic controller (6) is implemented. Using Matlab's Robust Control Toolbox [7] to compute (6), the performance index  $\gamma = 6.03$  is obtained and the control matrices are:

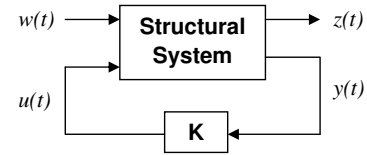


Fig. 6. Block diagram of simulink program for Quanser Shake Table II experiments.

$$A_k = \begin{bmatrix} -57.99 & -4.89 & -27.66 & 303.85 & -212.17 & 2.12 \\ 0.18 & -3.24 & -1.32 & 52.52 & 4.54 & -0.29 \\ -41.11 & -16.74 & -28.22 & 392.42 & -162.13 & -8.47 \\ 247.39 & 26.78 & 137.73 & -1655.66 & 967.55 & 4.66 \\ 245.41 & 49.72 & 129.82 & -1935.56 & 786.63 & 4.01 \\ 800.71 & 463.07 & 537.60 & -11120.36 & 2231.03 & -99.36 \end{bmatrix}$$

$$B_k = \begin{bmatrix} -0.09 & 1.90 & -0.28 \\ 0.06 & -0.29 & -0.19 \\ -2.87 & 2.50 & -3.33 \\ 14.54 & 1.55 & 16.70 \\ -32.72 & -9.90 & 17.81 \\ 24.16 & -160.28 & 76.91 \end{bmatrix}, \quad (12)$$

$$C_k = [13.13 \ 1.06 \ 8.41 \ -68.68 \ 72.10 \ -0.84].$$

#### B. Linear dynamic versus PD sample controls

To compare the performance of linear dynamic control (6) against the faulty structure case, we consider the sample controller provided in [17], a PD Position Control that implements the proportional-derivative feedback loop and calculates the motor input current needed to move the stage to the desired position. To simplify the presentation of figures, we will call it as Sample control, and the controller (6) with matrices (12) will be named after as Dynamic control. The two control strategies are compared in order to study the performance. In this paper, the external disturbance considered is a sine chirp wave (Quanser Chirp).

Figures 7 to 9 picture the time history plots under chirp input, showing open loop (uncontrolled case), dynamic and sample controls behavior. The accelerations of floor 1 and 2 are similar under both controllers (Figures 7 and 8) but the control performance of dynamic controller (6) is better than the sample controller. In Figure 9, where control performance is shown, we note that the control performance of dynamic control (6) improves the sample control [17], almost halving the control effort.

To compare more accurately the performance of the different control strategies, the  $H_\infty$  index is estimated as follows:

$$J_2 = \int_0^T \|u(t)\|_2^2 dt. \quad (13)$$

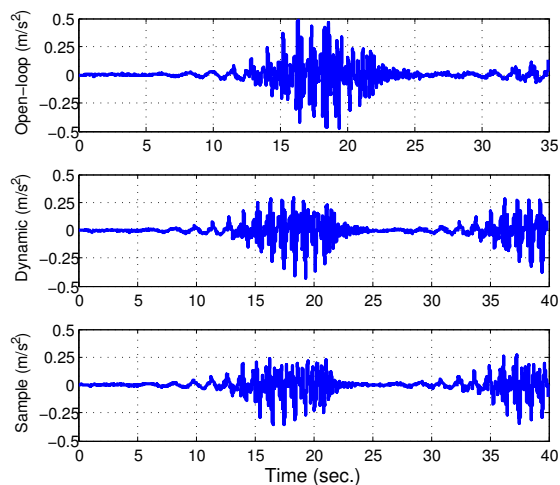


Fig. 7. Acceleration on Floor 1 under Chirp input.

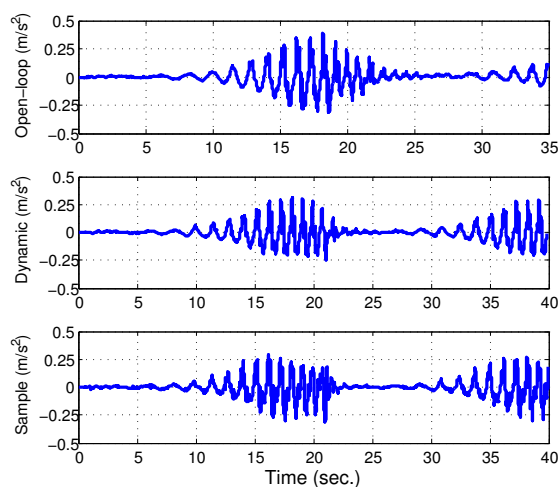


Fig. 8. Acceleration on Floor 2 under Chirp input.

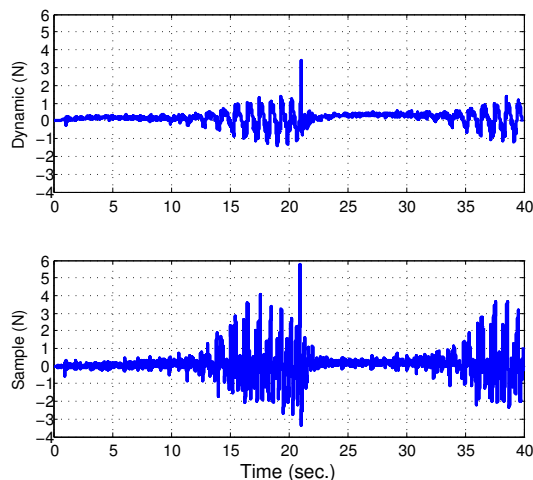


Fig. 9. Controls under Chirp input.

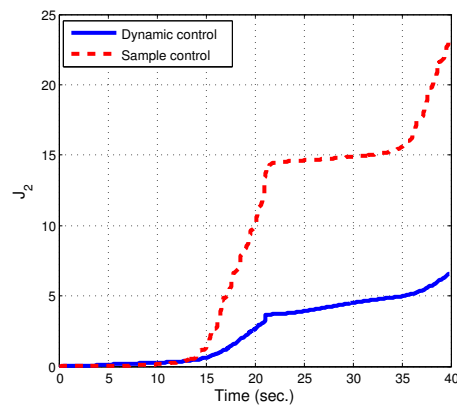


Fig. 10.  $H_\infty$  index  $J_2$  under Chirp input.

This index  $J_2$  is evaluated for all the time interval. Due to the numerical evaluation, the design has to be careful implemented on initial conditions.

Figure 10 pictures the  $H_\infty$  index  $J_2$  when a Quanser Chirp wave is introduced as perturbation. As Figure 10 pictures, the energy index  $J_2$  is worst in the sample case. In [4], an  $H_\infty$  is also designed for the same experiment. Note that the  $H_\infty$  weighting matrices induces a greater power consumption. In this sense, our control design is cheaper than [4].

#### IV. CONCLUSIONS AND FUTURE WORKS

##### A. Conclusions

In this paper, the design of an active controller that stabilizes a structural system was studied, in the presence of a faulty study case and external disturbances. A linear dynamic  $H_\infty$  control was developed for active mass damper systems subject to sinusoidal perturbation. This robust control was designed using linear matrix inequality (LMI) theory. According to experiments, where a flexible two levels building with active mass damper seismically excited was employed, it was appreciated a good performance of our linear dynamic design, when it is compared to a given PD controller, evidencing our control robustness. The experimental results, with external disturbance, demonstrated that the performance was satisfactory.

##### B. Future Works

To obtain a cheapest LMI control, a local controller will be designed, where only cart position can be used.

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